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| APPLICATION NO. | FILING DATE | FIRST NAMED INVENTOR | ATTORNEY DOCKET NO. | CONFIRMATION NO. | |
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| 09/900,771 | 07/06/2001 | Ichiro Mase | P/2856-22 | 7693 | |
| 2352 75 | 590 08/29/2003 | | | 10 | |
| OSTROLENK FABER GERB & SOFFEN | | | EXAMINER | | |
| | E OF THE AMERICAS NY 100368403 | AS | UHLIR, NI | UHLIR, NIKOLAS J | |
| · | • 5 • • | · · | ART UNIT | PAPER NUMBER | |
| | | | 1773 | | |
| | | DATE MAILED: 08/29/2003 | | | |

Please find below and/or attached an Office communication concerning this application or proceeding.

| | · . | | | // 1 | | | |
|---|---|--|--|--------------------|--|--|--|
| ٠ ' ' | * | Application No. | Applicant(s) | | | | |
| • | , | 09/900,771 | MASE ET AL. | | | | |
| | Office Action Summary | Examiner | Art Unit | | | | |
| | | Nikolas J. Uhlir | 1773 | • | | | |
| Period f | The MAILING DATE of this communication app or Reply | ears on the cover sheet with the | correspondence ad | dress | | | |
| A SH THE - Exte afte - If th - If No - Fail - Any | HORTENED STATUTORY PERIOD FOR REPLY MAILING DATE OF THIS COMMUNICATION. ensions of time may be available under the provisions of 37 CFR 1.13 r SIX (6) MONTHS from the mailing date of this communication. e period for reply specified above is less than thirty (30) days, a reply operiod for reply is specified above, the maximum statutory period vure to reply within the set or extended period for reply will, by statute, reply received by the Office later than three months after the mailing led patent term adjustment. See 37 CFR 1.704(b). | 36(a). In no event, however, may a reply be a within the statutory minimum of thirty (30) dayill apply and will expire SIX (6) MONTHS from the application to become ABANDON | timely filed ays will be considered timely m the mailing date of this co IED (35 U.S.C. § 133). | y. mmunication. | | | |
| 1)⊠ | Responsive to communication(s) filed on 05 2 | lune 2003 . | | . • | | | |
| 2a)□ | <u> </u> | is action is non-final. | | | | | |
| 3) Since this application is in condition for allowance except for formal matters, prosecution as to the merits is closed in accordance with the practice under <i>Ex parte Quayle</i> , 1935 C.D. 11, 453 O.G. 213. | | | | | | | |
| · _ | tion of Claims | | | | | | |
| 4)[4] | Claim(s) <u>1-20</u> is/are pending in the application | • | | | | | |
| € \□ | 4a) Of the above claim(s) is/are withdrawn from consideration. | | | | | | |
| | Claim(s) is/are allowed. | | | | | | |
| _ | ☐ Claim(s) 1-20 is/are rejected. | | | | | | |
| 7)∐ | • | r election requirement | | | | | |
| • | Claim(s) are subject to restriction and/o tion Papers | r election requirement. | | | | | |
| | The specification is objected to by the Examine | r. | | | | | |
| | The drawing(s) filed on is/are: a)□ accept | <u></u> | aminer. | | | | |
| Applicant may not request that any objection to the drawing(s) be held in abeyance. See 37 CFR 1.85(a). | | | | | | | |
| 11) The proposed drawing correction filed on is: a) approved b) disapproved by the Examiner. | | | | | | | |
| If approved, corrected drawings are required in reply to this Office action. | | | | | | | |
| 12) The oath or declaration is objected to by the Examiner. | | | | | | | |
| Priority | under 35 U.S.C. §§ 119 and 120 | | · | | | | |
| 13) Acknowledgment is made of a claim for foreign priority under 35 U.S.C. § 119(a)-(d) or (f). | | | | | | | |
| a) | a) ☐ All b) ☐ Some * c) ☐ None of: | | | | | | |
| | 1. Certified copies of the priority documents have been received. | | | | | | |
| | 2. Certified copies of the priority documents have been received in Application No | | | | | | |
| *: | Copies of the certified copies of the prior application from the International Bu See the attached detailed Office action for a list | reau (PCT Rule 17.2(a)). | | Stage | | | |
| 14) 🗌 . | Acknowledgment is made of a claim for domesti | c priority under 35 U.S.C. § 119 | e) (to a provisional | application). | | | |
| a) ☐ The translation of the foreign language provisional application has been received. 15) ☐ Acknowledgment is made of a claim for domestic priority under 35 U.S.C. §§ 120 and/or 121. | | | | | | | |
| Attachmei | nt(s) | | | | | | |
| 2) 🔲 Noti | ce of References Cited (PTO-892) ce of Draftsperson's Patent Drawing Review (PTO-948) rmation Disclosure Statement(s) (PTO-1449) Paper No(s) | 5) Notice of Informa | ary (PTO-413) Paper No Il Patent Application (PT | | | | |

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DETAILED ACTION

Claim Rejections - 5 USC § 103

- 1. The following is a quotation of 35 U.S.C. 103(a) which forms the basis for all obviousness rejections set forth in this Office action:
 - (a) A patent may not be obtained though the invention is not identically disclosed or described as set forth in section 102 of this title, if the differences between the subject matter sought to be patented and the prior art are such that the subject matter as a whole would have been obvious at the time the invention was made to a person having ordinary skill in the art to which said subject matter pertains. Patentability shall not be negatived by the manner in which the invention was made.
- 2. Claims 1-2, 4-5, 8-9, 12-13, 15-16, and 18-20 are rejected under 35 U.S.C. 103(a) as being unpatentable over Long et al. (US6176453) in view of Okamoto et al. (US2001/0027856).
- 3. Claim 1 requires a composite material heat controller for an object, the composite material heat controller comprising: a base material that radiates a larger amount of heat at high temperature relative to that of a low temperature, the base material having a surface adapted to thermally contact a surface of the object; and a phase change substance overlying the base material, wherein the phase change substance has insulation properties at a high temperature, metallic properties at low temperature, and the phase change substance radiates larger amounts of heat at high temperature relative to the amount of heat at low temperature, wherein the phase change substance has a high reflectivity in the thermal infrared region at low temperature.
- 4. Regarding these limitations, Long et al. (Long) teaches a radiator structure comprising a heat source 36 (equivalent to applicants claimed object), a radiator element 30 having an inner surface in thermal contact with the heat source through a thermal transfer medium 38, and a coating 44 in contact with the outer surface of the

radiator element (see figure 2a and column 4, lines 17-50). It is the examiners position that the radiator body 28 and the coating 44 are equivalent to the applicants claimed base material that radiates larger amounts of heat at high temperature then at low temperature, wherein the base material has a surface that is adapted to thermally contact the object.

- 5. Long fails to teach a phase change substance overlying the base material, wherein the phase change substance exhibits the properties required by claim 1.
- 6. However, regarding this deficiency, Okamoto et al. (Okamoto) teaches a heat control device suitable for use on an artificial satellite or spacecraft (page 1, section 1). This heat control device comprises a variable phase substance arranged on the heat radiation surfaces of a spacecraft. The variable-phase substance is a manganese perovskite oxide that undergoes a phase transition around room temperature. This substance has the characteristics of a metal at the low temperature phase, and the characteristics of an insulator at the high temperature phase. Further, this substance has a low heat radiation ratio at low temperature, and a high heat radiation ratio at high temperature (page 1, section 0016). Figure 2 clearly shows that this material exhibits higher infrared reflectivity in the low temperature phase as opposed to the high temperature phase. Thus, the phase change material of Okamoto meets the material property requirements of claim 1 for the required phase change material. This phasechange material is mounted in the form of a film on the heat radiation surfaces of a spacecraft, and so is lightweight and space saving (page 1-2, sections 19). Furthermore, this material regulates the amount of heat radiated from the surfaces of

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the spacecraft on order to control the internal temperature of the spacecraft (page 1, section 2 and section 15).

- 7. Therefore it would have been obvious to one of ordinary skill in the art at the time the invention was made to coat the phase change coating of Okamoto onto the surface of the coated radiator panel taught by Long.
- 8. One would have been motivated to make such a modification due to the teaching in Okamoto et al. that coating the radiator panel of a satellite with a phase change material of a manganese perovskite oxide allows the internal temperature of a spacecraft to be passively controlled within a desired temperature range.
- 9. Claim 2 requires the phase change substance to comprise a thickness in the range of about 1 to about 30μ. The examiner acknowledges that neither Long nor Okamoto teach this limitation. The examiner further acknowledges that Okamoto teaches in a specific example that the phase change material is suitably a several hundred micron thick film (page 2, section 22). However, the recitation of a single suitable thickness by Okamoto does not teach away from using a film having any other thickness. Bearing this in mind, it is noted that the phase change film substance of Okamoto is configured so as to form a *lightweight* heat control device. Given that Okamoto is concerned with weight, the examiner notes that it is well established that a thicker film of a substance necessarily weighs more than a thinner film of the same substance. Thus, the examiner takes the position that the thickness of the film of Okamoto is a results effective variable.

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10. Therefore it would have been obvious to one of ordinary skill in the art at the time the invention was made to control the thickness of the phase change film utilized by Long as modified by Okamoto to a desired range so as to obtain a coating having a desired weight.

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- 11. Claims 4 and 5 require the phase change substance to be a perovskite oxide (claim 4), more specifically manganese perovskite oxide (claim 5). These limitations are met as set forth above for claim 1.
- 12. Claim 8 requires a reflective plate or reflective film each having reflectivity with respect to visible light to be laminated onto the phase change substance on a side opposite the side on which the base material is laminated. Regarding this limitation, Okamoto et al. teaches that when the phase change material is mounted on a position that receives sunlight, a silicon plate that is transparent to thermal infrared but opaque to sunlight is positioned in front of the variable phase substance in order to minimize sunlight absorption (pages 1-2, section 19). As this silicon plate is opaque to visible light and is designed to minimize the absorption of sunlight, it is the examiners position that is will necessarily be reflective to visible radiation to some degree, and thus meets the reflection requirement in claim 8.
- 13. Therefore it would have been obvious to one with ordinary skill in the art at the time the invention was made to use the silicon plate taught by Okamoto et al. above the phase change material utilized by Long as modified by Okamoto et al.

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14. One would have been motivated to make this modification due to the teaching in Okamoto et al. that the silicon plate minimizes the absorption of sunlight by a phase change material that is mounted on the surface of a satellite that is exposed to sunlight.

- 15. Claim 9 requires a surface of the base material to be affixed to a surface of the object either directly or through an intervening heat conductive substance. With respect to this limitation, Long teaches that the radiator panel is attached to the heat source via a thermal transfer medium 38. It is the examiners position that this thermal transfer medium is equivalent to applicants claimed, "heat conductive substance."
- 16. Claim 12 requires the object in claim 1 to include a circuit used in a space vehicle, including man-made satellites and spaceships. This limitation is met as set forth above for claim 1, as Long and Okamoto clearly are directed towards the management of heat on spacecraft such as satellites.
- 17. Claims 13, 15-16, and 18-20 require a generic method for controlling heat in an object, wherein the method merely requires "attaching," or "providing" the layers required in claims 1, 4-5, 8-9 and 12. As the combination of Long with Okamoto necessarily requires these steps, the limitations of claims 13, 15-16 and 18-20 are met as set forth above for claims 1, 4-5, 8-9 and 12.
- 18. Claims 3, 6-7, 14, and 17 are rejected under 35 U.S.C. 103(a) as being unpatentable over Long as modified by Okamoto as applied to claims 1 and 13 above, and further in view of Babel et al. (US5296285).

- 19. Long as modified by Okamoto does not teach a composite material heat controller wherein the base material of claim 1 comprises a thickness greater then that of the phase change substance.
- 20. However, it is noted that in a specific example, Okamoto teaches that the phase change film is "several hundred microns thick" (page 2, section 22). The examiner interprets "several hundred microns thick to mean ≥200µ. It is well known in the art that the thickness of any solid layer is a results effective variable, with thicker layers weighing more than thinner layers. Thus, it would have been obvious to one with ordinary skill in the art at the time the invention was made to form the Okamoto et al. coating to the minimum acceptable thickness of 200µ in light of the fact that the Okamoto et al. coating is designed to be light weight and the minimum thickness of the specifically referenced range is "several hundred microns."
- 21. Regarding the thickness of the base material, it is noted that Long teaches that a suitable radiator element comprises an aluminum or aluminum alloy body that has been coated with a layer of white thermal control paint (column 4, lines 18-55). Long does not disclose a suitable thickness for the radiator element and white paint coating. Bearing this in mind, Babel et al. (Babel) teaches a high emittance, low absorptance coating for an aluminum substrate comprising a layer of anodized aluminum on the substrate, and a layer of white paint on the anodized aluminum (column 2, line 63-column 3, line 2). This coating is used as a thermal control surface of a spacecraft (column 4, lines 54-59). Babel teaches that the thickness of the anodized aluminum substrate and the white paint coating is in the range of 1.5-8 mils (38-203μ) (column 4, lines 44-53). Further,

Babel et al. teaches that the total thickness of the anodized aluminum and the high emissivity coating affects the corrosion resistance of the coating, with corrosion resistance increasing as the total thickness increases from 1-8 mils $(38-203\mu)$ (column 3, lines 10-28).

- 22. Therefore it would have been obvious to one of ordinary skill in the art at the time the invention was made to utilize the anodized aluminum substrate coated with a layer of white paint taught by Babel as the radiator element in Long.
- 23. One would have been motivated to make this modification in light of the fact that Long teaches that an aluminum or aluminum alloy substrate coated with a layer of white paint is suitable for use as the radiator element, and the fact that the anodized aluminum substrate coated with white paint taught by Babel is specifically taught to be useable for this exact purpose.
- 24. Further, given the fact that the Babel teaches that the total thickness of the anodized aluminum and the high emissivity coating affects the corrosion resistance, with corrosion resistance increasing as the total thickness increases from 1.5-8 mils (38-203μ) (column 3, lines 10-28), the examiner takes the position that the thickness of the anodized aluminum substrate and white paint coating is a results effective variable.
- 25. Therefore it would have been obvious to one of ordinary skill in the art at the time the invention was made to control the thickness of the anodized aluminum substrate and white paint coating to a desired range in order to achieve a desired level of corrosion resistance.

- 26. The limitations of claim 3 are met when the phase change substance is formed to as thickness of 200μ and the anodized aluminum/white paint radiator panel is formed to a thickness of 203μ .
- 27. Claim 6 requires the base material to have a thickness of 10-100μ. This limitation is met as set forth above for claim 3, when the thickness of the anodized aluminum/white paint radiator is controlled to thickness of 1.5mils (38μ).
- 28. Claim 7 requires the base material to include a material selected from the group consisting of silicone, alumina, and partially stabilized zirconia. This limitation is met as set forth above for claim 3, as anodized aluminum is known to have the formula Al₂O₃ which is also known in the art as alumina.
- 29. Claims 14 and 17 require a generic method of controlling heat in an object that requires "providing" or "forming" layers having the same limitations as claims 3 and 7. These limitations are met as set forth above for claims 3 and 7.
- 30. Claim 10 is rejected under 35 U.S.C. 103(a) as being unpatentable over Long as modified by Okamoto as applied to claim 1 above, and further in view of Bjorndahl et al. (US6005771).
- 31. Long as modified by Okamoto fails to teach the limitations of claim 10, wherein the applicant requires the base of claim 1 to be attached to the object via an appropriate intervening adhesive.
- 32. However, it is noted that Long teaches that the radiator panel is attached to the heat source via a thermal transfer medium, wherein the thermal transfer medium includes heat pipes, metallic strips, or other medium (column 4, lines 44-50).

33. Bearing the above in mind, Bjorndahl teaches that conduction of heat between a heat source (circuit) and a radiator panel can be improved by placing thermally conductive adhesive between the radiator panel and the heat source (column 1, lines 38-50).

- 34. Therefore it would have been obvious to one of ordinary skill in the art at the time the invention was made to utilizes thermally conductive adhesive as taught by Bjorndahl between the heat source and the radiator panel of Long as modified by Okamoto.
- 35. One would have been motivated to make this modification in order to enhance the conduction of heat between the heat source and the radiator panel.
- 36. Claim 11 is rejected under 35 U.S.C. 103(a) as being unpatentable over Long as modified by Okamoto as applied to claim 1 above, and further in view of Dalby (US4669685).
- 37. Long as modified by Okamoto does not teach the requirements of claim 11, wherein the applicant requires the object to comprise a non-flat surface.
- 38. For the purpose of this examination the examiner interprets "object comprises a non flat surface" to require an object generating heat to have at least 1 non-flat/planar surface. The examiner does not interpret this claim to require that the base material and phase change material be curved. Thus, claim 11 is read on by a satellite having heat generating elements incorporating curved fins, wherein a radiator panel is in thermal contact with the heat generating element and a phase change substance exhibiting the properties recited in claim 1 is applied to the radiator panel surface opposite the heat generating elements.

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39. Bearing the above interpretation in mind, Dalby teaches that the transfer of heat between heat generating elements and a heat radiator panel in a satellite is improved through the use of curved fins on the heat generating elements. Specifically, the use of curved fins allows heat generated from the heat producing elements to have a clear path to the heat radiator panel surrounding the heat-generating element (column 5, lines 35-50).

- 40. Therefore it would have been obvious to one of ordinary skill in the art to incorporate curved fins as taught by Dalby onto the heat generating elements taught by Long as modified by Okamoto.
- 41. One would have been motivated to make this modification in light of the teaching in Dalby that the transfer of heat between a heat generating element and a radiator panel in a satellite is improved by providing curved fins on the heat generating elements.

Response to Arguments

- 42. While the arguments dated 6/5/03 are rendered largely moot by the new grounds of rejection, for the purpose of clarity and to expedite the examination of the case, the examiner will address arguments that are still pertinent.
- 43. The only argument presented by the applicant that is still pertinent to the above rejection is the submission that the layer of paint taught by Long does not meet the limitations of the applicants claimed base material, as the paint coatings cannot exist independently of the object to which they are applied. Applicants assert that the ability of the base to exist independently of the object is required in method claim 13, which

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requires the base to be attached to the object. The examiner agrees with the applicants assertion that the base must be able to exist independently of the object to meet the applicants claim 13 limitations. While this argument is persuasive in overcoming the *prior* applied rejections utilizing Long and Okamoto, this argument does not overcome the new grounds of rejection, as the examiner now interprets the radiating panel 30 *and* the white paint coating 44 to meet the applicants claimed base requirement. Long clearly establishes that the radiating panel is a layer of aluminum or aluminum alloy, which can certainly exist as a separate entity from a supporting object. Thus, this argument is not persuasive in overcoming the applied prior art.

Conclusion

Any inquiry concerning this communication or earlier communications from the examiner should be directed to Nikolas J. Uhlir whose telephone number is 703-305-0179. The examiner can normally be reached on Mon-Fri 7:30 am - 5 pm.

If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Paul Thibodeau can be reached on 703-308-2367. The fax phone number for the organization where this application or proceeding is assigned is (703) 872-9306.

Any inquiry of a general nature or relating to the status of this application or proceeding should be directed to the receptionist whose telephone number is 703-305-0389.

N)U

Nju

Paul Thibodeau

Supervisory Patent Examine

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